# **Current status of power system for unmanned aerial vehicle fuel cell under different scenarios**

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**Abstract.** With the increasingly wide application of the unmanned aerial vehicle, a new power resource is in urgent demand in the market. Domestic and foreign researchers have also done much research on how to improve the power system. Considering the energy density, gravimetric specific power, economic benefits and environmental protection, fuel cell becomes the best choice to solve the problem of unmanned aerial vehicle endurance. The gravimetric specific power of fuel cells is low, and it is difficult to cope with the large range fluctuation of load power when the UAV performs complex flight tasks. After comparing the performance and products of various types of fuel cells, the cathode open air-cooled proton exchange membrane fuel cell is chosen as the most suitable one for UAVs. This paper will illustrate how fuel cell and battery hybrid power system works and introduce their structure. At the end of the paper, some conclusions and prospects for the development of UAV power systems are also made.

Keywords: unmanned aerial vehicle, fuel cell, power system.

#### 1. Introduction

The unmanned aerial vehicle (UAV) is operated by RF remote control and corresponding control programs. Compared with manned aircraft, its advantages are small and flexible, low manufacturing cost, easy to use, low environmental requirements and strong battlefield viability, etc. Today, with the increasingly wide application of UAVs, a new power structure is urgently needed in the market to replace the traditional fuel engine and lithium battery. Considering the energy density, gravimetric specific power, economic benefits and environmental protection, fuel cell becomes the best choice to solve the problem of UAV endurance. The gravimetric specific power of fuel cells is low, and it is difficult to cope with the large range fluctuation of load power when the UAV performs complex flight tasks. At present, fuel cells and other kinds of power sources are generally combined to form a hybrid power system to solve this problem. This paper focuses on fuel cell and battery hybrid power systems.

Although fuel cell has many advantages as the power source of UAVs, it also has the main problem of low power density. Researchers at home and abroad have done much research on how to improve the hybrid power system based on fuel cells [1].

This paper mainly introduces the composition of the fuel cell power system of UAVs. Then, analyze the advantages and disadvantages of current advanced cases in the industry. In the end, put forward prospects for future development.

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# 2. Composition of fuel cell power system for UAV

The power system of a fuel cell UAV is mainly composed of a hydrogen storage device, fuel cell, electronic governor, motor, propeller and other auxiliary power electronic equipment (Figure 1). Fuel cells and auxiliary power supply are the energy sources of UAVs, which are directly related to the important indicators of UAVs, such as endurance, flight distance and maximum load weight [2].

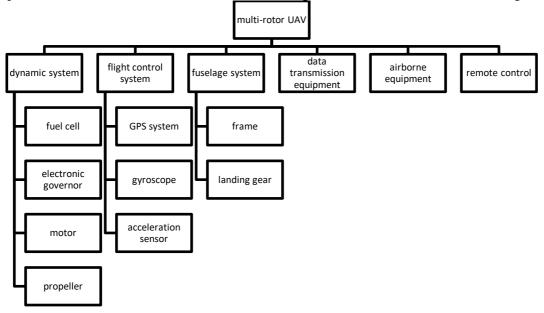


Figure 1. Multi-Rotor unmanned aerial vehicle system composition.

# 2.1. Operating principle of fuel cell

The UAV needs a power system with high energy density and light volume, which enables the UAV to carry more equipment, achieve more functions and extend its endurance. The fuel cell can directly convert the energy stored in the fuel into electric energy without driving the motor through fuel combustion, thus simplifying the power system structure.

Fuel cells are divided into different types according to the different electrolytes inside. The following chart shows the performance comparison of several main fuel cells.

Туре	Proton exchange membrane fuel cell	Solid oxide fuel cell	Direct methanol fuel cell
Type of fuel	Hydrogen	Hydrogen or hydrocarbons such as methane, propane, etc.	Methanol
Product	Water	Water	Water, carbon dioxide
Working temperature(°C)	30-110	600-1000	30-90
Power density of reactor(W/kg)	>500	>800	>70
System Power Density (W/kg)	>150	>100	>50

 Table 1. Performance comparison of different fuel cells.

Table 1 shows that the operating temperature of proton exchange membrane fuel cells is lower than that of other types of fuel cells. And its product, water, is environmentally friendly, making the proton exchange membrane fuel cells the most suitable for unmanned power systems.

The operating principle of a proton exchange membrane fuel cell is shown in Figure 2. Hydrogen is decomposed into hydrogen ions and electrons at the anode when the fuel cell is operating. The chemical formula for the anodic reaction is:

$$H_2 \to 2H^+ + 2e^- \tag{1}$$

Oxygen and electrons react with hydrogen ions at the cathode to produce water. The chemical formula for the cathodic chemical reaction is:

$$2H^{+} + \frac{1}{2}O_{2} + 2e^{-} \to H_{2}O$$
<sup>(2)</sup>

The total chemical reaction equation is:

$$H_2 + \frac{l}{2}O_2 \to H_2O \tag{3}$$

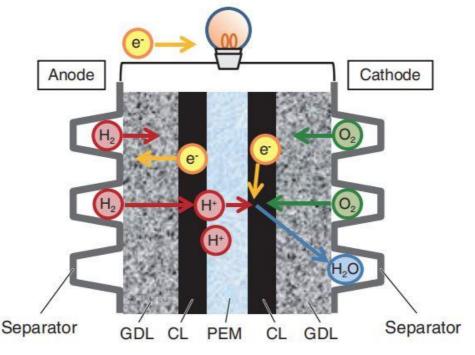


Figure 2. Operating principle of fuel cell

#### 2.2. Structure of fuel cells

Fuel cells are classified into air-cooled, liquid-cooled and many other types. Air-cooled type can be divided into cathode open type, and cathode closed type. The cathode open fuel cell has a relatively simple structure and high gravimetric specific power, suitable for UAV application as a power supply [3].

The operating voltage of the proton exchange membrane fuel cell (PEMFC) is usually between 0.5V and 0.7V, which is lower than required in most practical applications. Therefore, several cell units are usually used in series to form PEMFC stacks [4]. The reactor structure of open cathode fuel cells combines the above two structures, which means that while air passes through the cooling reactor from the reactor, it is used directly to provide oxygen for the reactions of the cathodes of each battery in the reactor. This structure reduces the mass and volume of the whole power system greatly based on output power, so it is suitable for use as the power supply for UAVs.

The cathode open fuel cell stack can be divided into two typical structures: the air self-breathing structure and the forced air convection structure with passive intake, as shown in Figure 3.

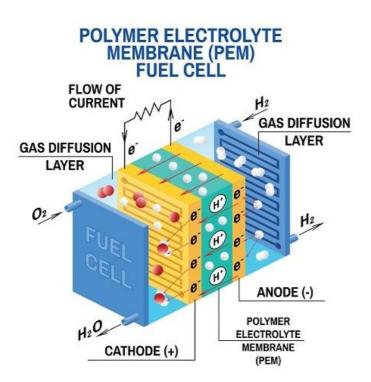


Figure 3. Structure schematic of open cathode fuel cell.

Compared with common fuel cell systems used on the ground, fuel cell systems for unmanned aerial vehicles have slightly different structures due to changes in temperature, air pressure, etc., due to changes in altitude [5]. Figure 4 shows the system structure of a fuel cell at a high altitude. It is mainly divided into four parts: fuel cell stack, hydrogen and oxygen storage supply system, control system and load system.

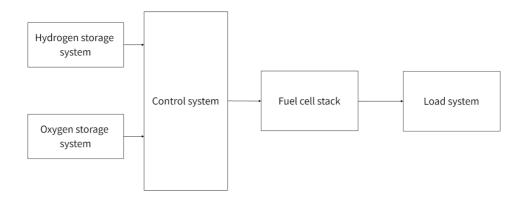


Figure 4. Unmanned aerial vehicle fuel cell system structure.

## 2.3. Hydrogen storage structure

The main type of hydrogen fuel used for UAVs is gaseous hydrogen. At present, a high-pressure cylinder is the most widely used hydrogen storage equipment. According to material and hydrogen storage pressure, gas cylinders can be divided into 4 categories [6]. As shown in Table 2, gas hydrogen storage pressure is generally 20-70 MPa. The maximum pressure of the hydrogen storage unit made of stainless steel is only 20 MPa, with the hydrogen storage ratio not exceeding 3%. It is generally used as an industrial hydrogen storage cylinder. Hydrogen storage with carbon fiber winding and high-density

polymer lining can withstand the pressure of 70 MPa and a hydrogen storage ratio of 11.3%, which is suitable for unmanned aerial vehicles.

Туре	Material	Maximum pressure / MPa	Hydrogen storage ratio / %
Ι	Stainless steel	20	1~3
II	Composite winding on steel/Aluminium substrates	30	
III	Composites (FRP, high- strength synthetic fiber or carbon fiber)	44	
IV	Carbon fiber wound high density polymer lining	70	11. 3

Table 2. Types	of high-pressure	hydrogen stora	ge bottles [7].

#### 2.4. Auxiliary power supply

There is a wide range of auxiliary power options in hybrid power systems. Lithium batteries and supercapacitors are common types of auxiliary power.

Lithium batteries have high energy storage density, long service life and up to 10,000 charge and discharge cycles. They usually form battery power packs in series to meet the needs of different voltage levels. The self-discharge rate of lithium batteries is as low as 1%, and their weight is about 1/5 of that of lead acid products at the same volume. With the development of battery technology, lithium batteries are widely used in new energy vehicles, energy storage devices and hybrid power systems. The relationship between voltage and SOC of a typical lithium battery is shown in Figure 5.

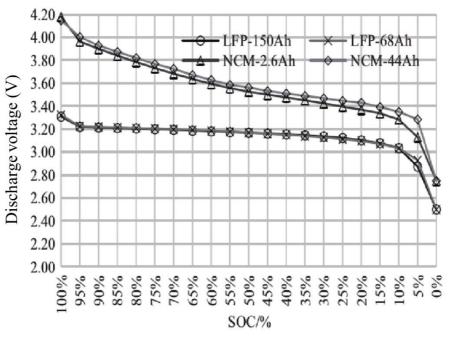


Figure 5. Change diagram of lithium battery voltage and SOC [8].

Ultracapacitor generally refers to double-layer capacitance, a new type of energy storage device. Ultracapacitors are between batteries and capacitors, and their enormous capacity can be used as batteries. From the perspective of practicality, ultracapacitors have short charging times and long service life. At the same time, it can work stably under various temperature conditions without harm to the environment. In the research and development process in related fields, lithium-ion supercapacitor has become the industry's preferred energy density with a relatively large increase compared with general double-layer capacitance. Typical lithium-ion ultracapacitor parameters are shown in Table 3. **Table 3.** Lithium-ion capacitor parameters [9].

	T T T T T T T T T T T T T T T T T T T					
Enterprise	Voltage window/V	Internal resistance/ mΩ	Energy density/(W·h/kg	Gravimetric specific power/(Kw/kg)		
JM Energy	2.2-3.8	1.2	10	21.5		
JM Energy	2.2-3.8	6.9	24	4.7		
TAIYO	2.2-3.8	-50AC	10			
YUDEN						
AFEC	2.0-4.0	1.5	10	17.2		
SHINKOBE-	2.2-3.8	3.5	8	7.4		
DENKI						
IEECAS	2.0-4.0	2.0	30	20		

#### **3. Industry case studies**

Following the application of fuel cells in different sizes of UAVs, Doosan Corporation of Korea, Proonex, AMI, EnergyOr, and HES in Europe and America, Troowin, and Ear in China started to develop fuel cells for light and small UAVs. AMI mainly develops solid oxide fuel cells, while other companies develop proton exchange membrane fuel cells. Some basic parameters are shown in Table 4. **Table 4.** Parameters of commercial fuel cells for light and small unmanned aerial vehicles [7].

Model	Manufacturer	Cooling mode	Output power/W	Mass/kg	Specific power/W· $kg^{-1}$
200W	Protonex	Liquid	200	2.7	74
500W	Protonex	Liquid	500	2.0	280
550W	Protonex	Liquid	550	1.9	289
600W	Protonex	Liquid	600	1.8	333
1200W	Protonex	Liquid	1 200	4.0	300
D245	AMI	SOFC	245	2.5	96.1
310-XLE	EnergyOr	Air	310	3.9	78.5
210-XLE	EnergyOr	Air	210	3.6	57.5
410-LE	EnergyOr	Air	410	3.4	120.6
A-200	HES	Air	200	0.7	310
A-500	HES	Air	1 500	1.4	360
A-1000	HES	Air	1 000	2.3	440
H1	MMC	Air	1 800	5.2	340
EOS600	Pearl	Air	600	3	200
EOS1000	Pearl	Air	1 000	4	250
HyLite1200	Troowin	Air	1 200	4	300

Company	Name	Battery	Power/W	Time of flight
Duralite Powe	DP-A-3000	PEMFC		2
AMI		SOFC	60	4.3
Georgia		PEMFC	550	0.75
Institute of				
Technology				
EnergyOr	H2Quad	PEMFC		2
United States	Ion Tiger	PEMFC		48
Naval Research	C			
Laboratory				
H2GO				1.5
DOOSAN	DT20	PEMFC	2100	2
Intelligent		PEMFC		12
Energy				

Table 5. Current research status of fuel cell unmanned aerial vehicle abroad [10].

Due to the characteristics of the fuel cell, such as high-power generation efficiency, portability, environmental protection and low power density, domestic and foreign researchers have conducted several studies on unmanned aerial vehicles with the hybrid power system (Table 5). Gong A and others compared the performance of a fuel cell/battery hybrid UAV under different mission scenarios. The comparison results show that the selection of a UAV mission profile through fuel cell performance is conducive to reducing hydrogen consumption during UAV flight. Kim T. et al. designed and built a fuel cell/hybrid battery system UAV, which consists of a fuel cell stack and a hydrogen generator, and conducted flight tests. The UAV successfully flew for two hours [10]. In 2019, MetaVista Corporation of Korea, in conjunction with Intelligent Energy, a British fuel cell engineering company, developed a four-wing fuel cell unmanned aircraft (Figure 6). It uses an ultra-light liquid hydrogen storage tank that provides 1860 watt-hours per kilogram of energy density for 11 hours of flight time, breaking the world record for multirotor unmanned aerial vehicles.



## Figure 6. MetaVista UAV.

On January 23, 2022, the first flight of the Green Gull 30, a Hydrogen-powered UAV independently developed by the Research Center for Hydrogen Power and Low Carbon Energy in Chongqing Research Institute of Harbin Institute of Technology, was successfully completed. The maximum take-off weight of the Green gull 30 is 30kg, and its wingspan of it is 4m. Its cruise speed is 18-25ms<sup>2</sup>. The cruise time is 9h, the load is 3kg, and the range is 800km. The endurance of this UAV is about 3-4 times higher than that of the lithium battery UAV of the same power level. The original impingement ventilation fuel cell power system was used on this UAV, which reduces power loss due to the additional load of the fuel cell system and greatly improves the operating efficiency of the system. The Green gull 30 is now the

fixed-wing UAV with the longest endurance in China. It can be applied to scenes such as long-term investigation, line patrol, aerial surveys, material transportation and fire warning.

# 4. Conclusion

Through the above research, the most suitable power system for UAV should be a hybrid power supply consisting of proton exchange membrane fuel cell and lithium-ion battery or ultracapacitor, supplemented by carbon fiber winding and high-density polymer lined hydrogen storage system. After years of research and development, the development of fuel cell/battery hybrid UAVs has rapidly moved from new concept exploration and key technology research to practicality and engineering. In the future, the potential of fuel cell energy density should be fully exploited, and the lightweight of UAV power system should be further studied to maximize the endurance time; Carry out research on the overall design method of aerodynamic, structural, energy and power integration to improve the payload ratio. Integrating auxiliary energy sources such as lithium batteries and ultracapacitors to optimize overall system power output, improve economic benefits and expand application scope.

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